

U.S.N.

B.M.S. College of Engineering, Bengaluru-560019

Autonomous Institute Affiliated to VTU

June 2025 Semester End Main Examinations

Programme: B.E.

Branch: Aerospace Engineering

Course Code: 23AS3PCASD

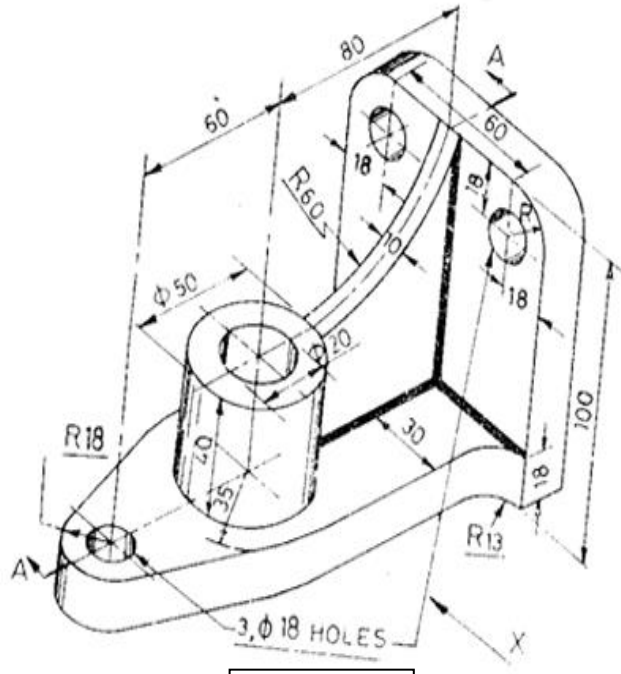
Course: Aerospace Drafting

Semester: III

Duration: 3 hrs.

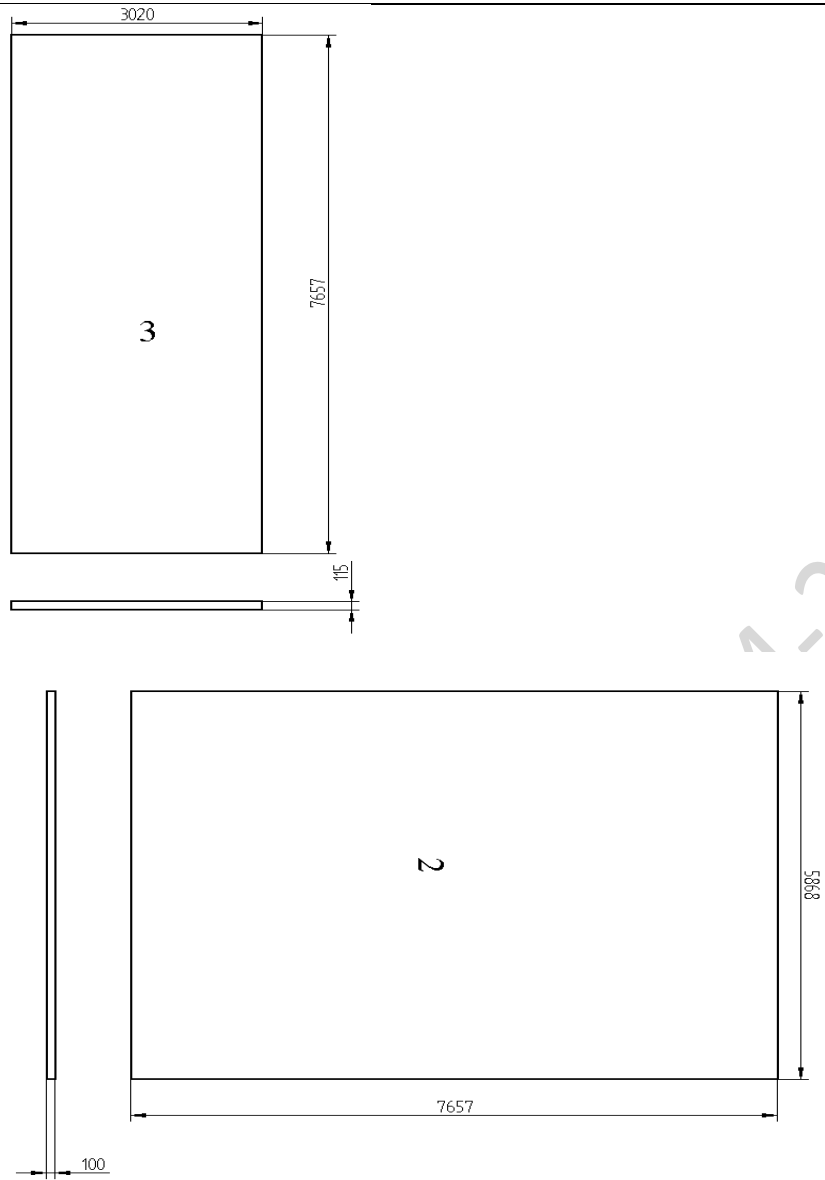
Max Marks: 100

- Instructions:** 1. Answer any three full questions, choosing one full question from each part.
2. Missing data, if any, may be suitably assumed.

Important Note: Completing your answers, compulsorily draw diagonal cross lines on the remaining blank pages. Revealing of identification, appeal to evaluator will be treated as malpractice.			PART -A	CO	PO	Marks
	1		For the object shown (FIG.Q1) below draw the three views. Show all dimensions.  FIG.Q1	CO1, CO2	PO2	20
			OR			
	2		Draw the profile of the following threads and indicate all the dimensions. <ol style="list-style-type: none"> ISO thread (Internal and External) having pitch 60mm, ACME thread having pitch 50mm. 	CO1, CO2	PO2	20

		PART-B																																					
3		Draw to 1:1 scale the top and sectional front views of a double riveted lap joint with zig –zag riveting. The thickness of the plates is 9 mm. Show at least three rivets in each row. Indicate all the dimensions. Use snap headed rivets.	CO1, CO2	PO2	20																																		
		OR																																					
4		Draw to 1:1 scale the top and sectional front views of a double riveted butt joint with double cover plates and chain riveting. The thickness of the plates is 9 mm. Show at least three rivets in each row. Indicate all the dimensions. Use snap headed rivets.	CO1, CO2	PO2	20																																		
		PART-C																																					
5		<p>The details of the propeller and hub assembly are shown in the FIG.Q5. Draw front, top and left views of the assembly. Use suitable scale.</p> <p style="text-align: center;"><u>Propeller and Hub</u></p> <p>Thickness of the blade (in mm)</p> <table><tr><th>Height of The HUB</th><th>HUB</th><th>A-A</th><th>B-B</th><th>C-C</th><th>D-D</th><th>E-E</th></tr><tr><td rowspan="2">34</td><td>7</td><td>7</td><td>7</td><td>7</td><td>4</td><td>2</td></tr><tr><td>CHORD</td><td>CHORD</td><td>CHORD</td><td>CHORD</td><td>CHORD</td><td>CHORD</td></tr><tr><td></td><td>5</td><td>5</td><td>8</td><td>8</td><td>3</td><td>1</td></tr><tr><td></td><td>Width</td><td>34</td><td>38</td><td>34</td><td>20</td><td>10</td></tr></table> <p>1. Mount Plate 2. Propeller 3. Face plate 4. Lock Bolt</p>	Height of The HUB	HUB	A-A	B-B	C-C	D-D	E-E	34	7	7	7	7	4	2	CHORD	CHORD	CHORD	CHORD	CHORD	CHORD		5	5	8	8	3	1		Width	34	38	34	20	10	CO3	PO3	60
Height of The HUB	HUB	A-A	B-B	C-C	D-D	E-E																																	
34	7	7	7	7	4	2																																	
	CHORD	CHORD	CHORD	CHORD	CHORD	CHORD																																	
	5	5	8	8	3	1																																	
	Width	34	38	34	20	10																																	

			OR			
	6		<p>The Details of fuselage is shown in the fig. Q6. Draw front, top and left views of the assembly.</p>	CO3	PO3	60



Item Number	Title	Material	Quantity
1	CASING	Aluminum, 6061-T6	1
2	CENTRE FLOORING	Aluminum, 6061-T6	1
3	LOWER FLOORING	Aluminum, 6061-T6	1

Fig.Q6
